# **Optimization Procedure for Identifying Constitutive Properties of High Speed Induction Motor**



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#### **Objectives**

In order to predict the rotordynamics of a high speed induction motor in bending, an optimization procedure is proposed for identifying the constitutive properties especially those of the magnetic core. Modal parameters predicted by a finite element model based mainly on beam elements, and measured on an induction motor are included in modal error functions contained in a functional. The minimization of this functional by using the Levenberg-Marquardt algorithm permits extracting the constitutive properties **along** the magnetic core.

## **Squirrel Cage Rotor**

- Assembly Structure:
  - → Tie rods, Short-circuiting rods, Laminated stack.
- Prestressed Structure.
- Laminations without central hole.

#### **Magnetic Core Modeling**

- Short-circuiting rods are modeled as beams, of diameter  $D_{CC}$  (Fig. 3), whose neutral axes coincide with the neutral axis of the magnetic core.
- Tie rods can be seen as <u>external tendons</u>.
- Laminated stack is modeled as a <u>distributed</u> orthotropic material.

# **Beam Finite Element Model**

Tie rods stiffness matrix at clamping nodes  $A_0$  and  $B_0$  (Fig. 2):  $K_{TI} = \beta \cdot \frac{E_{TI} S_{TI}}{L_{TI}} \cdot e_{TI}^{2} \cdot \begin{vmatrix} 0 & 1 \\ 0 & 0 \end{vmatrix}$ 





# **Constitutive Property Distributions**

Distributions  $x \in \Re^n$ , n = 3h,  $h \in \mathbb{N}$ , considered by splitting the stack into *h* sub-domains  $\Omega_r$  such as  $(x_p)_{p=3r-2,3r-1,3r} = \{E_r, G_r, v_r\}, r = 1, ..., h.$ 



### **Modal Error Functions**

The <u>difference</u> between measured and predicted natural
The <u>difference</u> is defined as a non-linear least squares frequencies and mode shapes is quantified by four modal error functional, a global error norm: functions, for <u>each mode k, k = 1...m</u>:

 $F_{k}^{\omega}\left(\mathbf{x}^{i}\right) = \frac{\omega_{k}}{\omega_{k}} - 1, \quad F_{k}^{\varphi_{ED}}\left(\mathbf{x}^{i}\right) = \sum_{j=1, j \neq k}^{m} \left|MAC\left(\tilde{\varphi}_{j}^{i}, \varphi_{k}\right) - MAC\left(\varphi_{j}, \varphi_{k}\right)\right|, \quad f\left(x^{i}\right) = \frac{1}{2} \left\|F^{Tot}\left(x^{i}\right)\right\|^{2} = \frac{1}{2} \sum_{k=1}^{q} \left[F_{k}^{Tot}\left(x^{i}\right)\right]^{2}, q = 4 \times m.$  $F_{k}^{\varphi_{D}}\left(x^{i}\right) = 1 - MAC\left(\widetilde{\varphi}_{k}^{i}, \varphi_{k}\right), \quad F_{k}^{\varphi_{EC}}\left(x^{i}\right) = \sum_{j=1}^{r} \left| \frac{\left(\widetilde{\varphi}_{k}^{i}\right)_{j}}{\max\left(\left(\widetilde{\varphi}_{k}^{i}\right)_{j}\right)} - \frac{\left(\varphi_{k}\right)_{j}}{\max\left(\left(\varphi_{k}\right)_{l}\right)} \right| \right|$ **Multi-Objective Function**  $F^{Tot}(x^{i}) = \left[\alpha^{\omega}F^{\omega}, \alpha^{\varphi_{D}}F^{\varphi_{D}}, \alpha^{\varphi_{HD}}F^{\varphi_{HD}}, \alpha^{\varphi_{Ec}}F^{\varphi_{Ec}}\right]^{T},$ 

#### Levenberg-Marquardt Algorithm

Optimization strategy consists in <u>minimizing</u> the <u>difference</u> between the measured and predicted <u>modal</u> data, at each iteration *i*:

 $\left[x^{0},\lambda_{0}\right]$  given

 $\left\{ d_{i} = -\left(J\left(x^{i}\right)^{T} J\left(x^{i}\right) + \lambda_{i}I\right)^{-1} \cdot \nabla f\left(x^{i}\right), \text{ with } \nabla f\left(x^{i}\right) = J\left(x^{i}\right)^{T} F^{Tot}\left(x^{i}\right). \right\}$  $x^{i+1} = x^i + \rho_i d_i$ 

#### **Undamped Eigenvalue Problem**

 $\begin{cases} (K - \lambda_k M) \widetilde{\varphi}_k = 0 \\ \widetilde{\varphi}_k^T M \widetilde{\varphi}_k = 1 \end{cases}, \ \lambda_k = \widetilde{\omega}_k^2, \ k = 1, \dots, m, \end{cases}$ 

where *K* and *M* are the stiffness and mass matrices respectively.

#### **Analytical Eigen-Derivatives**





→ Weight coefficients based on the initial mean value, *i = 0*.



1≠K Efficient Convergence.

Less CPU-time Consuming.

The *k*<sup>th</sup> eigenvector derivative requires only the k<sup>th</sup> eigenvector.

Figure 9. Normalized Coulomb's modulus distributions.

0.3

Abscissa (m)

# **Industrial Application**



Figure 5. Experimental Setup.

#### **Method Accuracy**

 $\rightarrow$  Modal Testing  $\{\varphi_k, \omega_k\}, k = 1, \dots, 4.$ 

The global error norm decreases with the number of sub-domains.

The global error norm obtained with h = 7is <u>height times lower</u> (Fig. 6) than this one obtained with an isotropic assumption.

The proposed method is more <u>accurate</u> identification classical methods than considering restrictive assumptions such as homogenous or isotropic material.

Table 1 First four bending measured and predicted natural frequencies obtained with 7 sub-domains.

	1 <sup>st</sup>	2 <sup>nd</sup>	3 <sup>rd</sup>	4 <sup>th</sup>
Measured (Hz)	856	1230	1713	2330
Predicted (Hz)	847	1263	1705	2315
Error (%)	1.05	2.68	0.47	0.64

#### **Sub-Domain Dependence**

• Young's and Coulomb's modulus distributions tend to <u>stabilize</u> if h > 7 sub-domains.

Distributions have <u>constant</u> values in the <u>middle</u> (55 % of the magnetic core length) of the magnetic core and <u>decrease</u> at its <u>ends</u>.



Figure 8. Normalized Young's modulus distributions.

#### Conclusions

Optimization procedure tested on an industrial rotor with <u>complex design</u>. Constitutive property distributions <u>depend</u> on the <u>number of sub-domains</u>



Figure 6. The dashed and solid line represent measured and Figure 7. Evolution of the global error norm versus the number of predicted mode shapes respectively. Predicted mode shapes were sub-domains. obtained with 7 sub-domains.

![](_page_0_Picture_60.jpeg)

used in the finite element (FE) modeling of the magnetic core.

Distributions tend toward a particular shape versus the number of subdomains: <u>Constant</u> values in the middle (55% of the magnetic core length) and Low values at its ends.

The low level permits probably to consider the shear effect located close to the ends.

This procedure is very useful for establishing an FE model based mainly on beam elements and therefore containing few degrees of freedom.

→ Great advantage for predicting the rotordynamics, *i.e.* unbalance responses and transient responses.

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